

# 7 Secular Perturbations

- Terms independent of  $\lambda$  and  $\lambda'$  → long-term effect of perturber's gravity
- Gauss' averaging method: sec pots are equivalent to pot = from potential obtained by spreading the perturber's mass around its orbit w line density determined by velocity  
→ think of pl system as interacting wires.

## Secular disturbing function from MD99 Appendix B

All direct terms, zeroth order, from Table B.1, w  $j=0$ ; so k 2nd order:

$$R_0^{sec} = [f_1 + f_2(e^2 + e'^2) + f_3(s^2 + s'^2)] \cos(\varpi) + f_{10} ee' \cos(\varpi' - \varpi) + f_{14} ss' \cos(\Omega' - \Omega) + 4^{th} \text{ order terms}$$

where  $s = \sin i/2 \rightarrow I/2$

NOTE: terms involving eccentricity and incl are decoupled to 2nd order (as  $\varpi = 0(s^2, s'^2, ss')$ )

Getting constants from Table B.3:

$$R_0^{sec} = \frac{1}{2} b_{1/2}^0(\alpha) + \frac{1}{8} (2\alpha D + \alpha^2 D^2) b_{1/2}^0(\kappa) (e^2 + e'^2) - \frac{1}{4} \alpha (b_{3/2}^{-1}(\kappa) + b_{3/2}^1(\kappa)) [(I/2)^2 + (I'/2)^2] + \dots$$

$$+ \frac{1}{4} [2 - 2\alpha D - \alpha^2 D^2] b_{1/2}^0(\kappa) ee' \cos(\varpi' - \varpi) + \alpha b_{3/2}^2(\kappa) (I/2)(I'/2) \cos(\Omega' - \Omega)$$

where  $\alpha = a/a'$

Simplifying Laplace coefficients [B.2]:

$$b_s^j(\alpha) = \frac{1}{\pi} \int_0^{2\pi} \cos^j \psi [1 - 2\alpha \cos \psi + \alpha^2]^{-s} d\psi$$

$$D b_s^j(\alpha) = \partial/\partial \alpha (b_s^j(\alpha)) = \frac{1}{\pi} \int_0^{2\pi} \cos^j \psi (2s \cos \psi - 2\alpha s) [1 - 2\alpha \cos \psi + \alpha^2]^{-s-1} d\psi$$

Using  $2 \cos \psi \cos^j \psi = \cos(j-1)\psi + \cos(j+1)\psi$

$$D b_s^j(\alpha) = \frac{1}{\pi} \int_0^{2\pi} [s \cos(j-1)\psi + s \cos(j+1)\psi - 2\alpha s \cos^j \psi] [1 - 2\alpha \cos \psi + \alpha^2]^{-s-1} d\psi$$

$$= s b_{s+1}^{j-1}(\alpha) + s b_{s+1}^{j+1}(\alpha) - 2\alpha s b_{s+1}^j(\alpha)$$

etc

$$(2\alpha D + \alpha^2 D^2) b_{1/2}^0(\alpha) = \alpha b_{3/2}^1(\kappa)$$

$$(2 - 2\alpha D - \alpha^2 D^2) b_{1/2}^0(\alpha) = -\alpha b_{3/2}^2(\kappa)$$

$$\text{So } R_0^{sec} = \frac{1}{2} b_{1/2}^0(\alpha) + \frac{1}{8} \alpha b_{3/2}^1(\kappa) [e^2 + e'^2] - \frac{1}{8} \alpha b_{3/2}^1(\kappa) [I^2 + I'^2] - \frac{1}{4} \alpha b_{3/2}^2(\kappa) ee' \cos(\varpi' - \varpi) + \frac{1}{4} \alpha b_{3/2}^2(\kappa) II' \cos(\Omega' - \Omega)$$

AND  $R = (\mu/a') R_0^{sec}$

$R' = (\mu/a) R_0^{sec}$

Rewrite using new variables:  $k = e \cos \varpi, h = e \sin \varpi$   
 $q = I \cos \Omega, p = I \sin \Omega$  } [7.3]

$e^2 = k^2 + h^2$

and  $ee' \cos(\varpi' - \varpi) = ee' [\cos \varpi \cos \varpi' + \sin \varpi \sin \varpi'] = kk' + hh'$  etc.

Ignoring  $\frac{1}{2} b_{1/2}^0(\alpha)$  term:

$$R_0^{sec} = \frac{1}{8} \alpha b_{3/2}^1(\kappa) [h^2 + k^2 + h'^2 + k'^2 - p^2 - q^2 - p'^2 - q'^2] - \frac{1}{4} \alpha b_{3/2}^2(\kappa) [kk' + hh'] + \frac{1}{4} \alpha b_{3/2}^2(\kappa) [qq' + pp']$$

## Consider a system of $N$ planets secularly interacting planets

Potential is additive, so for the  $j$ -th planet, noting that only terms involving elements of that planet are important.

$$R_j = \sum_{i=1, i \neq j}^{N-1} \left( \frac{G m_i}{a_{out}} \right) \left[ \frac{1}{8} \left( \frac{a_{in}}{a_{out}} \right) b_{3/2}^1 \left( \frac{a_{in}}{a_{out}} \right) [k_j^2 + h_j^2 - q_j^2 - p_j^2] - \frac{1}{4} \left( \frac{a_{in}}{a_{out}} \right) b_{3/2}^2 \left( \frac{a_{in}}{a_{out}} \right) [k_j k_i + h_j h_i] + \frac{1}{4} \left( \frac{a_{in}}{a_{out}} \right) b_{3/2}^2 \left( \frac{a_{in}}{a_{out}} \right) [q_j q_i + p_j p_i] \right]$$

Rewrite more succinctly:

$$R_j = n_j a_j^2 \left[ \frac{1}{2} A_{jj} (k_j^2 + h_j^2) + \frac{1}{2} B_{jj} (q_j^2 + p_j^2) + \sum_{i=1, i \neq j}^{N-1} [A_{ji} (k_j k_i + h_j h_i) + B_{ji} (q_j q_i + p_j p_i)] \right]$$

where  $A_{ji} = -\frac{1}{4} n_j \left( \frac{m_i}{m_*} \right) \alpha_{ji} \bar{\alpha}_{ji} b_{3/2}^2(\alpha_{ji}) = \left( \frac{G m_i}{a_{out}} \right) \left( -\frac{1}{4} \frac{a_{in}}{a_{out}} b_{3/2}^2 \left( \frac{a_{in}}{a_{out}} \right) \right) / n_j a_j^2$

$B_{ji} = \frac{1}{4} n_j \left( \frac{m_i}{m_*} \right) \alpha_{ji} \bar{\alpha}_{ji} b_{3/2}^2(\alpha_{ji})$

$A_{jj} = -B_{jj} = \sum_{i=1, i \neq j}^{N-1} B_{ji}$

$\alpha_{ji} = a_i/a_j, \bar{\alpha}_{ji} = 1$  if  $a_j > a_i$   
 $\alpha_{ji} = a_j/a_i, \bar{\alpha}_{ji} = a_j/a_i$  if  $a_j < a_i$  } ie  $\alpha_{ji} = a_{in}/a_{out}, \bar{\alpha}_{ji} = a_j/a_{out}$

NB  $G m_i/a_{out} = G m_i \left( \frac{M_i}{M_*} \right) / a_{out} = n_j^2 a_j^2 \left( \frac{M_i}{M_*} \right) \bar{\alpha}_{ji}$

Get eqns of motion from Lagrange's Planetary Equations

To lowest order, from (7.3):

$$\begin{aligned} \dot{e} &= -(na^2e)^{-1} \partial R / \partial \omega \\ \dot{\omega} &= (na^2e)^{-1} \partial R / \partial e \\ \dot{I} &= -(na^2I)^{-1} \partial R / \partial \Omega \\ \dot{\Omega} &= (na^2I)^{-1} \partial R / \partial I \end{aligned}$$

Thus from (7.3):

$$\begin{aligned} \dot{h} &= (\partial h / \partial e) \dot{e} + (\partial h / \partial \omega) \dot{\omega} = (na^2e)^{-1} [ -(\partial h / \partial e) [ (\partial R / \partial e) (\partial k / \partial \omega) + (\partial R / \partial \omega) (\partial h / \partial e) ] + (\partial h / \partial \omega) [ (\partial R / \partial e) (\partial k / \partial e) + (\partial R / \partial e) (\partial h / \partial e) ] ] \\ &= (na^2e)^{-1} [ -\sin \omega (-e \sin \omega) + e \cos \omega \cos \omega ] \partial R / \partial k \\ &= (na^2)^{-1} \partial R / \partial k \end{aligned}$$

cancels

cancels

And similarly  $\dot{k} = -(na^2)^{-1} \partial R / \partial h$   
 $\dot{q} = -(na^2)^{-1} \partial R / \partial p$   
 $\dot{p} = (na^2)^{-1} \partial R / \partial q$

(7.6)

Applying this to (7.5):

$$\begin{aligned} \dot{h}_j &= A_{jj} k_j + \sum_{i=1, i \neq j}^{Np} A_{ji} k_i \\ \dot{k}_j &= -A_{jj} h_j - \sum_{i=1, i \neq j}^{Np} A_{ji} h_i \\ \dot{p}_j &= B_{jj} q_j + \sum_{i=1, i \neq j}^{Np} B_{ji} q_i \\ \dot{q}_j &= -B_{jj} p_j - \sum_{i=1, i \neq j}^{Np} B_{ji} p_i \end{aligned}$$

(7.7)

$= \sum_{i=1}^{Np} A_{ji} k_i$

Can write even more succinctly using  $\underline{z} = \begin{pmatrix} z_1 e^{i\omega_1 t} \\ \vdots \\ z_{Np} e^{i\omega_{Np} t} \end{pmatrix} = \begin{pmatrix} k_1 + ih_1 \\ \vdots \\ k_{Np} + ih_{Np} \end{pmatrix}$  (7.8)

$\underline{y} = \begin{pmatrix} y_1 e^{i\Omega_1 t} \\ \vdots \\ y_{Np} e^{i\Omega_{Np} t} \end{pmatrix} = \begin{pmatrix} q_1 + ip_1 \\ \vdots \\ q_{Np} + ip_{Np} \end{pmatrix}$

As  $\dot{z}_j = k_j + ih_j = \sum_{i=1}^{Np} A_{ji} (-h_i + ik_i) = \sum_{i=1}^{Np} A_{ji} z_i$

$$\begin{aligned} \dot{\underline{z}} &= i A \underline{z} \\ \dot{\underline{y}} &= i B \underline{y} \end{aligned}$$

where  $A = \begin{pmatrix} A_{11} & A_{12} & \dots \\ A_{21} & & \\ \vdots & & A_{Np, Np} \end{pmatrix}$   $B = \begin{pmatrix} B_{11} & B_{12} & \dots \\ B_{21} & & \\ \vdots & & B_{Np, Np} \end{pmatrix}$  (7.9)

Solve as an eigenvalue problem

From (7.2):  $\underline{z} = W a e^{i\Lambda t} W a^{-1} z_0$ ,  $\underline{y} = W b e^{i\Lambda t} W b^{-1} y_0$

OR  $\underline{z} = \begin{pmatrix} e_1 & e_2 & \dots \\ e_{21} & & \\ \vdots & & \end{pmatrix} \begin{pmatrix} e^{i[\omega_1 t + \phi_1]} \\ e^{i[\omega_2 t + \phi_2]} \\ \vdots \\ e^{i[\omega_{Np} t + \phi_{Np}]} \end{pmatrix}$  and  $z_j = \sum_{i=1}^{Np} e_{ji} e^{i[\omega_i t + \phi_i]}$  (7.10)

$\underline{y} = \begin{pmatrix} I_1 & I_2 & \dots \\ I_{21} & & \\ \vdots & & I_{Np, Np} \end{pmatrix} \begin{pmatrix} e^{i[\Omega_1 t + \delta_1]} \\ e^{i[\Omega_2 t + \delta_2]} \\ \vdots \\ e^{i[\Omega_{Np} t + \delta_{Np}]} \end{pmatrix}$

- This is Laplace-Lagrange secular solution showing that orbital evolt is sum of Np sinusoidal oscills w periods given by eigenfrequencies (that are real)
- Good for  $e, I \ll 1$  (no orbits overlapping or near resonance); at higher order e and I no longer decoupled
- One of incl<sup>n</sup> eigenvalues is 0 because choice of ref plane is arbitrary and only mutual incl<sup>n</sup> is important (Eq 4.1); not so for  $\Omega$  as pericentre introduces a preferred orientation

\* Note that (7.2) was solution to  $\dot{X} = AX \rightarrow X = W a e^{i\Lambda t} W a^{-1} X_0$   
 working through the derivation w extra "i" leads to the given equation

# Test Particles in a Planetary System (Eq 4.2)

Are subject to same disturbing force Eq 4.1, which for the Np system before can be described by Eq 5, but dropping the subscript "j", also dropping i ≠ j condition in the sum (NOTE I swapped i → j to allow i to be used later):

$$\therefore R = na^2 \left[ \frac{1}{2} A (k^2 + h^2) + \frac{1}{2} B (l^2 + p^2) + \sum_{j=1}^{Np} [A_j (k k_j + h h_j) + B_j (q q_j + p p_j)] \right]$$

where  $A_j = -\frac{1}{4} n \left( \frac{M_j}{M_0} \right) \alpha_j \bar{\alpha}_j b_{3/2}^2(\alpha_j)$

$B_j = \frac{1}{4} n \left( \frac{M_j}{M_0} \right) \alpha_j \bar{\alpha}_j b_{3/2}^2(\alpha_j)$

$A = -B = \sum_{j=1}^{Np} B_j$

$\alpha_j = a_j/a, \bar{\alpha}_j = 1$  if  $a > a_j$

$\alpha_j = a/a_j, \bar{\alpha}_j = a/a_j$  if  $a < a_j$

Eq 11

Can we Lagrange's planetary eqns Eq 6, which gives an analogous set of eqns Eq 7 with dropped subscripts "j" etc:

$\dot{h} = Ak + \sum_{j=1}^{Np} A_j k_j$

$\dot{k} = -Ah - \sum_{j=1}^{Np} A_j h_j$

that can be combined using  $z = e e^{i\Omega t} = k + ih$

$\dot{z} = iAz + i \sum_{j=1}^{Np} A_j z_j$

where  $z_j = \sum_{i=1}^{Np} e_{ji} e^{i(g_i t + \beta_i)}$  from L-L sol = Eq 10

Eq 12

Solve this using integrating factor

$$\frac{d}{dt} [z e^{-iAt}] = i \sum_{j=1}^{Np} A_j z_j e^{-iAt}$$

$$= i \sum_{j=1}^{Np} A_j \sum_{i=1}^{Np} e_{ji} e^{i(g_i t + \beta_i - At)}$$

If  $A \neq A(t)$  then  $\int A dt = At + \beta$

$$z = e^{i(At + \beta)} \int \sum_{i=1}^{Np} \left( \sum_{j=1}^{Np} A_j e_{ji} \right) i e^{i(g_i - A)t + \beta_i - \beta} dt$$

Let  $v_i = \sum_{j=1}^{Np} A_j e_{ji}$

$$z = e_p e^{i(At + \beta)} + \sum_{i=1}^{Np} \left[ \frac{v_i}{g_i - A} \right] e^{i(g_i t + \beta_i)}$$

Eq 13

where  $e_p$  and  $\beta$  are constants of integration set by initial conditions

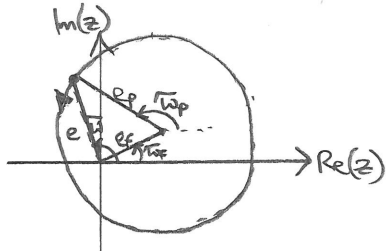
Write this as:

$$z = z_p + z_f = e_p e^{i\Omega t} + e_f e^{i\omega t}$$

Eq 14

where  $e_f, \omega$ : are forced eccentricity and forced longitude of pericentre that are known functions of  $(m_i, a_i, e_i, \omega_i)$  that vary slowly w sec. orb<sup>2</sup> of planet orbits.

$e_p, \Omega$ : are proper (or free) eccentricity and proper longitude of pericentre, set by initial condns (ie are intrinsic to particle), though  $\Omega$  increases linearly w time at rate A



So particle precesses anticlockwise around a circle of radius  $e_p$  at a rate A, where the centre of the circle also moves w time

## Inclination evolution

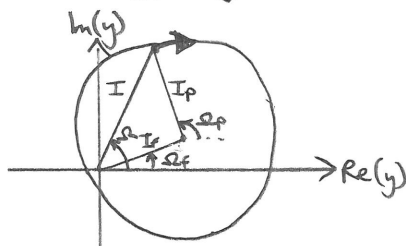
Similar:  $\dot{y} = -iAy + i \sum_{j=1}^{Np} B_j y_j$

$$y = I_p e^{i(-At + \gamma)} - \sum_{i=1}^{Np} \frac{M_i}{-A - f_i} e^{i(f_i t + \gamma_i)}$$

$$= y_p + y_f = I_p e^{i\Omega t} + I_f e^{i\omega t}$$

where  $M_i = \sum_{j=1}^{Np} B_j I_{ji}$

Eq 15



So particle precesses clockwise ...

Secular resonances

Locations where precession rate equals one of eigenfrequencies of planetary system

$A = g_i$  or  $-A = f_i$  7.16

at which point  $z_i \rightarrow \infty$  or  $y_i \rightarrow \infty$

Since  $g_i, f_i$  are independent of time, and  $A$  only depends on  $a$  (and  $m_i, a_i$  that are fixed), secular resonances are at fixed location in  $a$

- eg Asteroid belt is bounded by secular resonances (but are surfaces in  $a, e, I$  due to higher order terms)
- Secular resonances may move in time eg if  $q_i, M_i$  are varying due to non-secular processes
- secular resonance sweeping

Forced elements near a planet L

Now  $z_i = \sum_{j=1}^{N_i} \left[ \frac{\sum_{k=1}^{N_i} A_{jk} e_{jk}}{g_i - A} \right] e^{i(g_i t + \beta_i)}$

As  $\kappa_L = a/a_L \rightarrow 1, b_{3/2}^2(\kappa_L) \rightarrow \infty$  and  $b_{5/2}^2(\kappa_L) \rightarrow \infty$

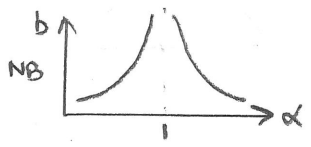
$\therefore A_L \rightarrow \infty$  and  $\sum_{j=1}^{N_i} A_{jk} e_{jk} \rightarrow A_L e_{Li}$

Also,  $A \rightarrow \infty$  and  $g_i - A \rightarrow -A = -B_L$

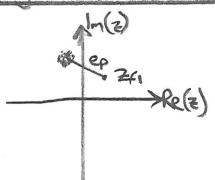
$\therefore z_i \rightarrow \sum_{j=1}^{N_i} \left( \frac{A_{jk}}{-B_L} \right) e_{jk} e^{i(g_i t + \beta_i)} = \left( \frac{A_L}{-B_L} \right) z_{Li} = \frac{b_{3/2}^2(\kappa \rightarrow 1)}{b_{5/2}^2(\kappa \rightarrow 1)} z_{Li} = z_{Li}$  7.17

$\therefore$  forced elements are planets orbit

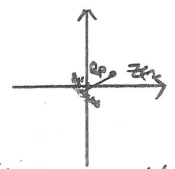
see MD99 eps 7.87, 88 for proof that this  $\Rightarrow 1$



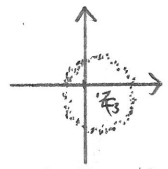
Hirayama asteroid families



i. Asteroid is  $e_p$  breaks into fragments



ii.  $z_i$  moves and fragments process at slightly different rates



iii. Fragments end up distributed around circle of radius  $e_p$  centred on forced elements

Expansions for high  $e, I$ :

Take additional terms in disturbing function, or use hierarchy if  $a_2 \gg a_1$  and expand in  $\kappa$  using Legendre...

- Quadrupole expansion is to  $O(\kappa^2)$   
eg. Kozai (1962) for  $M_0 \gg M_2 \gg M_1$  and  $e_2 = 0$
- Octupole expansion is to  $O(\kappa^3)$   
eg. Lee & Peale (2003) for coplanar system, Ford et al (2000)
- Higher orders, eg. Migaszewski & Wozniakowski (2008)

Secular terms are found by double averaging  $\langle\langle R \rangle\rangle = \left( \frac{1}{2\pi} \right)^2 \iint R dM dM'$

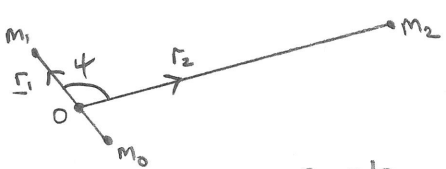
Secular perturbations with...

The disturbing function is additive, so can take eqs of motion 7.9 or 7.12 and add effects of other perturbations.

- eg Murray & Dermott showed the effect of resonant forces between two of the planets
- Ex 2010 Q3 considered effect of idleness.

Kozai Mechanism (see ch.9 of Valtonen & Karttunen)

For hierarchical systems, Jacobi coordinates are used to remove high order variation in orbital elements of  $m_2$  due to motion of  $m_1$  around  $M_0$  (EX2010Q1)

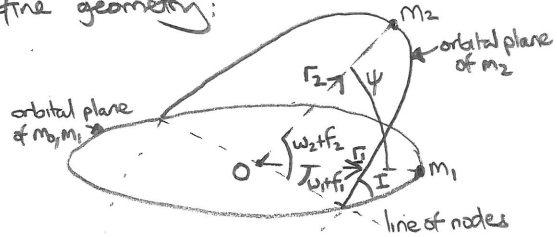


$r_1$  = pos of  $m_1$  wrt.  $M_0$   
 $r_2$  = pos of  $m_2$  w.r.t. c.o.m. of  $M_0$  and  $m_1$

To second order in expansion of  $r_1/r_2$ , the disturbing  $F$  is (ie quadrupole):

$$R = \frac{G M_0 m_1 m_2}{2(m_0 + m_1)} \left(\frac{r_1}{r_2}\right)^2 (3 \cos^2 \psi - 1) \quad \boxed{7.18}$$

Define geometry:



$$\cos \psi = \cos(\omega_2 + f_2) \cos(\omega_1 + f_1) + \sin(\omega_1 + f_1) \sin(\omega_2 + f_2) \cos I \quad \boxed{7.19}$$

Double averaging:  $\langle\langle R \rangle\rangle = \left[ \frac{G m_0 m_1 m_2 a_1^2}{8(m_0 + m_1) a_2^3 (1 - e_2^2)^{3/2}} \right] [2 + 3e_1^2 - 3 \sin^2 I (5e_1^2 \sin^2 \omega_1 + 1 - e_1^2)] \quad \boxed{7.20}$

Use Lagrange's planetary equations:

Outer orbit  $m_2$ :  $\dot{a}_2 = \dot{e}_2 = \dot{I}_2 = 0$  and only orientation of orbit vary (but not shape)

Inner orbit  $m_1$ :  $\dot{a}_1 = 0$  (by defn of sec. pert)

$$\begin{aligned} \dot{e}_1 &= \frac{15}{8} e_1 \sqrt{1 - e_1^2} \sin 2\omega_1 \sin^2 I \\ \dot{\omega}_1 &= \frac{3}{4} (1 - e_1^2)^{-1/2} [2(1 - e_1^2) + 5 \sin^2 \omega_1 (e_1^2 - \sin^2 I)] \\ \dot{I} &= -\frac{15}{8} e_1^2 (1 - e_1^2)^{-1/2} \sin 2\omega_1 \sin I \cos I \\ \dot{\Omega}_1 &= -\frac{1}{4} \cos I (1 - e_1^2)^{-1/2} [3 + 12e_1^2 - 15e_1^2 \cos^2 \omega_1] \end{aligned} \quad \boxed{7.21}$$

where unit of time is  $G m_0 m_2 [(m_0 + m_1) a_2^3 (1 - e_2^2)^{3/2}]^{-1}$

This means that  $\sqrt{1 - e_1^2} \cos I = \text{const} \quad \boxed{7.22}$

Can be solved, but for qualitative soln, consider evolt when  $e_1 \ll 1$  (ie ignore terms  $O(e_1^2)$ )

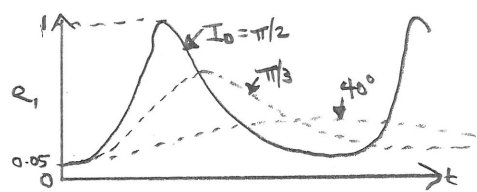
$$\begin{aligned} \dot{I} &\approx 0 \\ \dot{\omega}_1 &\approx \frac{3}{4} (2 - 5 \sin^2 I \sin^2 \omega_1) \end{aligned}$$

This can be solved leading to qualitatively different behaviour depending on  $I$  wrt  $I_{crit} = 39.2^\circ \quad \boxed{7.23}$

- For  $\sin^2 I < 0.4$ ,  $\dot{\omega}_1 > 0$  always, get sinusoidal oscillation in  $e_1$
- For  $\sin^2 I > 0.4$ , as  $t \rightarrow \infty$ ,  $\sin \omega_1 \rightarrow \sqrt{0.4} / \sin I$

$$\dot{e}_1 \propto e_1 \text{ s.t. } e_1 = 2 \cdot 10^{-6}$$

So  $e_1$  grows rapidly, but by  $\boxed{7.22}$   $I$  must decrease and there is a maximum eccentricity  $e_{max} = \sqrt{1 - \cos^2 I_0 / \cos^2 I_{min}}$  where  $\sin^2 I_{min} = 0.4 \quad \boxed{7.24}$



Kozai cycle where eccentricity (and inclination) oscillate with a period

$$P_{Kozai} \approx \left(\frac{a_2}{a_1}\right)^3 \left(\frac{m_0}{m_2}\right) t_{per} \quad \boxed{7.25}$$

$\boxed{eg}$  If  $I > 39.2^\circ$ , an outer binary would subject an inner planetary system to large increases in eccentricity, with correspondingly large changes in  $I$ , on the timescale given by  $\boxed{7.25}$

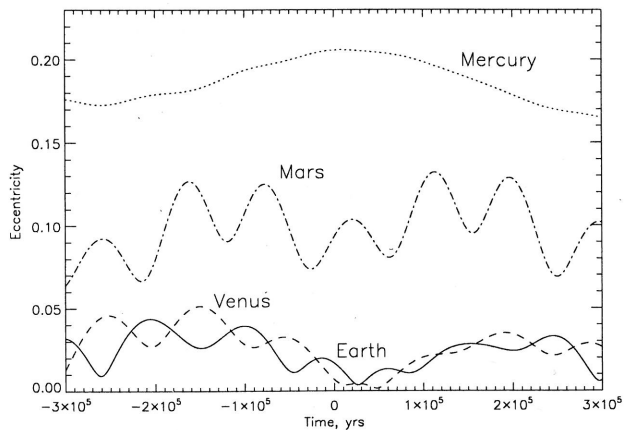
Although formally it does this regardless of  $m_2$ , the timescale is long for small  $m_2$ , and secular precession can be much smaller than that due to gravity of other planets etc, preventing this behaviour.

Another application is to circumplanetary orbits: the secular pert of Sun's gravity cause Kozai cycles in irregular satellites with  $I = 39^\circ - 41^\circ$

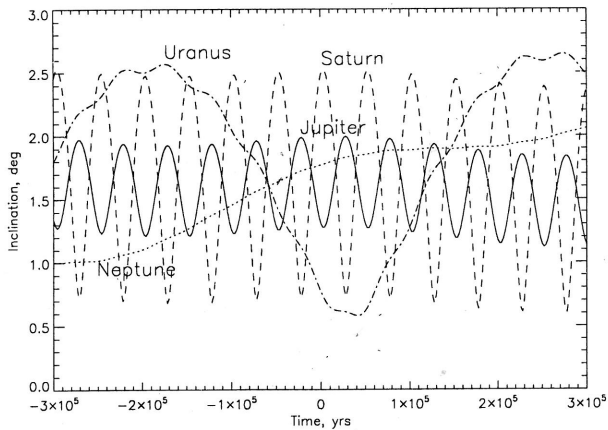
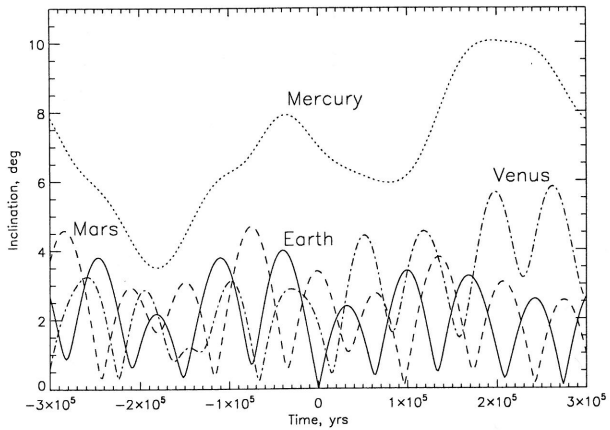
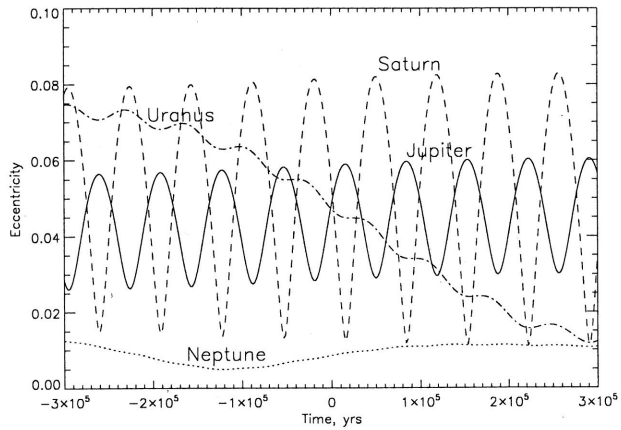
## Secular Perturbations

### Secular Evolution of Solar System Planet Orbits

Inner Solar System



Outer Solar System



**Secular Perturbations**  
Secular Evolution of Test Particles in the Solar System

